

T.D.3 - Forces Centrales.

Ex 1.

1) $E_p = -\frac{k m_1 m_2}{r} + \text{cte.}$

2) $\vec{F} = qvB\vec{u}_r \rightarrow E_p = \text{Non.}$

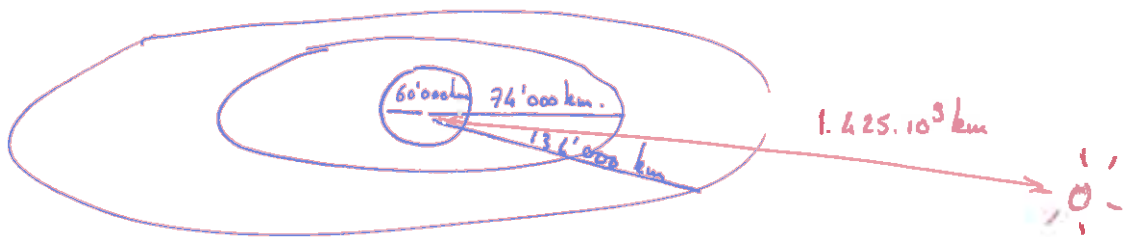
3) Non.

4) Non.

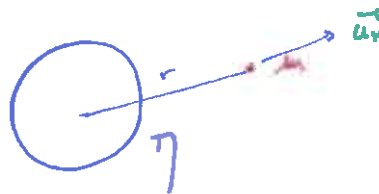
5) $\vec{z} \uparrow$

6) $\vec{F} = qvB\vec{u}_r \rightarrow E_p = -\frac{1}{2} qv^2 B + \text{cte}$

Ex 2



1) L'interaction de gravité =



$\vec{F} = -\frac{mM\eta G}{r^2} \vec{u}_r$

2) Satellite gravitant s/ une orbite circulaire =

⊙ Ref. saturno centrig. : centre = centre de Saturne.
axes = vers étoiles fixes.

⊙ $\vec{a} = -\frac{\eta G}{r^2} \vec{u}_r$
 $a = -\frac{v^2}{r} \vec{u}_r$ } $\rightarrow \text{vitesse } \sqrt{\frac{\eta G}{r}} = \text{cte.}$

⊙ Temps μ effectuer une orbite : $v = \frac{2\pi r}{T} \Rightarrow T = 2\pi r \sqrt{\frac{r}{\eta G}}$

↳ $T^2 = \frac{4\pi^2 r^3}{\eta G}$ 3^e loi de Kepler.
 ηG constante Kepler

$\eta = \frac{4\pi^2 r^3}{T^2 G} = \frac{4\pi^2 \times (159 \cdot 10^6)^3}{6.67 \cdot 10^{-11} \times (17 \times 3600 \cdot 58 \times 60)^2} = 5.7 \cdot 10^{26} \text{ kg}$

3) Dispositif d'une série d'objets punctuels s/ une \vec{u} orbitale.

Ils ont ts la \vec{u} vitesse \Rightarrow se déplacent collectivement.

4) Dispositif de 2 objets s/ 2 orbites \neq .

$$T = 2\pi r \sqrt{\frac{r}{\gamma G}} \Rightarrow r \uparrow T \uparrow \Rightarrow \text{A en retard s/B.}$$

Config. 2.

5) Au passage de Saturne.

$\vec{\eta} \perp \vec{r} \Rightarrow \vec{v} \perp \vec{u}$, si \vec{u} au passage.
Aucun aut vit. \neq



\Rightarrow vit. $\neq \Rightarrow$ f. de cisail^{te} \Rightarrow aurait de explorer.

$$\|\vec{v}\| = \sqrt{\frac{\gamma G}{r}} \uparrow qd r \downarrow$$

6) de l'opinion de Roche.

$$F_{AB} = \frac{m^2 G}{4\rho^2}$$

$$F_{S/A} = \frac{m\gamma G}{(r-p)^2}$$

$$F_{S/B} = \frac{m\gamma G}{(r+p)^2}$$

$$F_{S/A} - F_{S/B} = m\gamma G \left[\frac{1}{(r-p)^2} - \frac{1}{(r+p)^2} \right]$$

$$= m\gamma G \frac{r^2 - p^2}{r^2(r+p)^2}$$

$$F_{S/A} - F_{S/B} = 4m\gamma G \frac{p}{r^3}$$

car $\rho \ll r$
 $\frac{4rp}{(r-p)^2(r+p)^2} \approx \frac{4rp}{r^4}$

$$R_0 / \frac{m^2 G}{4\rho^2} \approx 4\gamma \frac{p}{R_0}$$

$$\Rightarrow R_0 = \sqrt[3]{\frac{4\gamma T}{\rho}}$$

Marius.

$r > R_0$ F_{AB} game
 \Rightarrow gros objet
 $r < R_0$ F_{AB} pend
 \Rightarrow s'annule.

Satellites → au-delà de R_0 .

Exercice 3 - Orbite de la lune 2 Terre.

1) $\vec{0}\vec{1} = r \vec{u}_r$

$\vec{L} = m \vec{r} \wedge r \dot{\theta} \vec{u}_\theta = m r^2 \dot{\theta} \vec{u}_z$

$\vec{v} = R \dot{\theta} \vec{u}_\theta$

$\vec{a} = -R \dot{\theta}^2 \vec{u}_r$

2) $E_c = \frac{1}{2} m R^2 \dot{\theta}^2$

$E_p = -\frac{\eta_T m G}{R}$

$E_m = \frac{1}{2} m R^2 \dot{\theta}^2 - \frac{\eta_T m G}{R}$

3) $E_m = \frac{L^2}{2m R^2} - \frac{\eta_T m G}{R} = \frac{G \eta_T m^2 R}{2 \mu R^2} - \frac{\eta_T m G}{R}$

$E_m = -\frac{3}{2} \frac{\eta_T m G}{R}$

4) $\mathcal{A} = \frac{L_0}{2m} T = \pi R^2$

$T = \frac{2\pi R^2}{\sqrt{G \eta_T m^2 R}} = \sqrt{\frac{4 \pi^2 R^3}{G \eta_T}} = T$

5) $\frac{T^2}{R^3} = \frac{4 \pi^2}{G \eta_T}$ indpt de m .

$E_m = \frac{1}{2} m R^2 \dot{\theta}^2 - \frac{m \eta_T G}{R}$

$L^2 = (m r^2 \dot{\theta})^2 = m^2 R^4 \dot{\theta}^2$

$L = m R^2 \dot{\theta}$

$E_c = \frac{1}{2} m R^2 \dot{\theta}^2$

$L^2 = m^2 R^4 \dot{\theta}^2$

$E_p = -\frac{\eta_T m G}{R}$

$E_m = \frac{1}{2} m R^2 \dot{\theta}^2 - \frac{m \eta_T G}{R}$

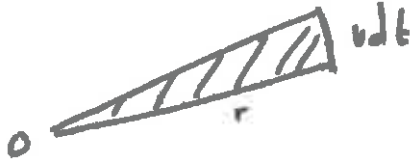
$\frac{L^2}{R} = m^2 R^3 \dot{\theta}^2$

$\Delta E_m = 0 \Rightarrow E_m = \text{cte}$

$E_m = -\frac{1}{2} G \frac{\eta_T m}{R} = -\frac{G \eta_T m}{2R} = E_c$

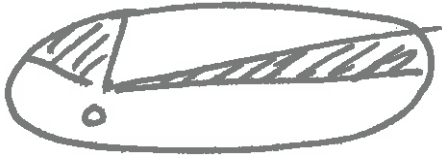
$E_c = \frac{1}{2} \frac{G \eta_T m}{R} = \frac{1}{2} G$

3. Ex d'applications.



- Aire balayée pendant dt.

$$dA = \frac{1}{2} r \omega dt = \frac{1}{2} r^2 \dot{\theta} dt = \frac{l_0}{2} \omega dt$$



d'air balayée ptt dt est tjrs la m^me = constante.

Problème: Conservat- de l'Em.

- Thm l'Em: $\Delta E_m = W_{f. non cons.} = 0$ en l'abs. d'autres forces.

- $E_m = \frac{1}{2} m v^2 + E_p = \text{cte.}$

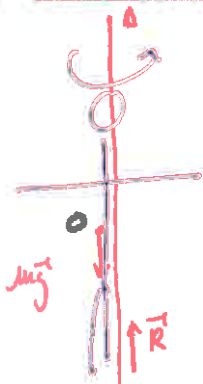
$$= \frac{1}{2} m (\underbrace{\dot{r}}_{\vec{e}_r} + \underbrace{r \dot{\theta}}_{\vec{e}_\theta})^2 = \frac{A}{r}$$

$$E_m = \frac{1}{2} m \dot{r}^2 + \underbrace{\frac{1}{2} m r^2 \dot{\theta}^2}_{E_{pot}} = \frac{A}{r}$$

E_{pot} pour étudier traj. possibles:

- paraboles
- hyperboles.
- cercles
- ellipses.

III Patineur.



$t_0 = 0$

$I_0 = \sum m_i r_i^2$



$t_1 > 0$

$I_1 = \sum m_i r_i^2$

$$\frac{d\vec{L}_0}{dt} = \vec{M}_{\vec{r},0} + \vec{M}_{\vec{F},0} = \vec{0}$$

$\vec{L}_0 = I \omega \vec{u}_3 = \text{cte}$

$I_0 \rightarrow I_1 < I_0$

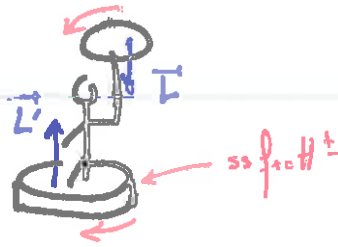
$\omega_0 \rightarrow \omega_1 > \omega_0$

La vitesse de rotati ↗

IV Gyroscope (1)

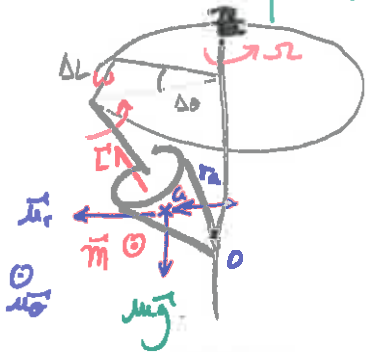


$t = 0$
 $L_s = 0$



t_1
 $\frac{d\vec{L}}{dt} = \vec{\sigma} \Rightarrow$ la chaise tourne de la direction opposée par compensation

V Toupie



La poid exerce un $\vec{M}_{P,0}$:

$$\vec{M}_{P,0} = r_a \vec{u}_r \wedge m\vec{g} = mgr_a \vec{u}_\theta = \frac{d\vec{L}}{dt}$$

- La toupie a un \vec{L}_s cinétique
 $\vec{L}_s = I\omega \vec{u}_z \quad \Delta$
(solide en rotation)

- La poid exerce un $\vec{M}_{P,0}$

$$\vec{M}_{P,0} = r_a \vec{u}_r \wedge m\vec{g} = mgr_a \sin\phi \vec{u}_\theta$$

$$\frac{d\vec{L}}{dt} = mgr_a \sin\phi \vec{u}_\theta$$

Or $\Delta L \Delta \theta L \sin\phi \Rightarrow \Delta \theta = \frac{\Delta L}{L \sin\phi} \Rightarrow \Omega = \frac{\Delta \theta}{\Delta t} = \frac{\Delta L}{\Delta t} \frac{1}{L \sin\phi}$

$\Omega = \frac{mgr_a}{I\omega}$

qd $r_a \nearrow$ (toupie tombe) $\Rightarrow \Omega \nearrow$.